

Heat Transfer Enhancement of a Two-Pass Cooling Channel for Gas Turbine Blades

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Introduction

It is crucial to perform high thermal efficiency in a gas turbine. Increasing the inlet temperature in the combustor is the way to achieve this goal for a gas turbine. Nonetheless, increasing the temperature gradient within the blade material, which causes thermal stresses. These effects must be reduced to achieve higher thermal efficiency. The two-pass cooling channels have been studied due to their performance in heat transfer. The dimpled surface on a pass wall cooling technology gets popular in recent years as it shows a low-pressure loss penalty. It is also advantageous to the turbine stages where lower pressure cooling is employed. This study is going to investigate turbulent flow and heat transfer in a two-pass duct. Three cases were considered here-

- Case A: Hemispherical Dimpled Channel
- Case B: Leaf Dimpled Channel
- Case C: Hemispherical Dimpled Channel without Bend

Methods

The Computational Fluid Dynamics (CFD) method was used for modeling the setup and analyzing the flow and heat transfer using Star CCM+ software. The turbulent flow field was conducted through the two-pass duct with $k-\omega$ SST model by the following literature [1,2,3]. The inlet is chosen larger than the outlet to balance the pressure on both sides.

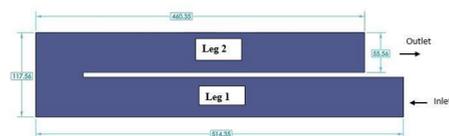


Figure 1: Two pass cooling channel (Front view, all dimensions in mm)

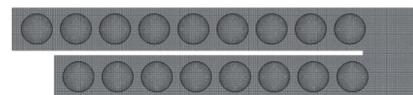


Figure 2: Hemispherical dimpled channel (Meshed scene, Bottom View)

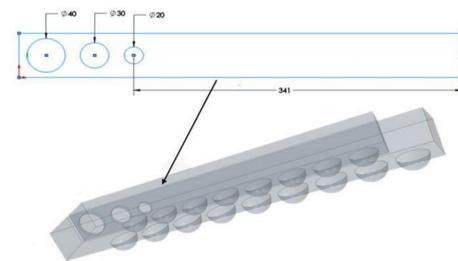


Figure 3: Two-pass cooling channel eliminating bend (all dimension in mm, Case C)

Table 1: Dimple Parameters

Name	Depth to Diameter ratio	Number of Dimples in Leg-1	Number of Dimples in Leg-2
Case A	0.5	9	8
Case B	0.5	9	8
Case C	0.5	9	8

Table 2: CFD Study Parameters

Model used	RANS ($k-\omega$ SST)
Number of cells	6.2 Million
Working fluid	Air
Inlet velocity	6.25 m/s
Reynolds number	20,000
Outlet	Pressure outlet
Heat flux (Bottom wall)	2,700 W/m ²

Results

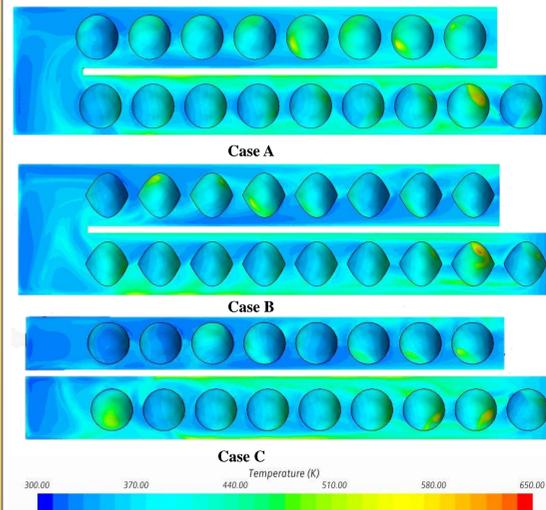


Figure 4: Temperature distribution along the channel at Reynolds number = 20,000

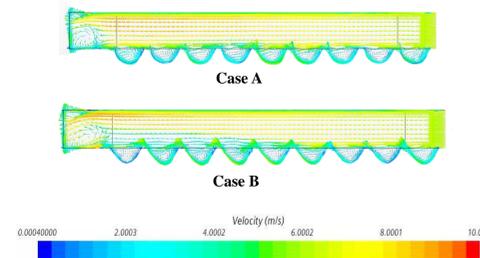


Figure 5: Velocity distribution along the leg-1 at Reynolds number = 20,000 (Side view)

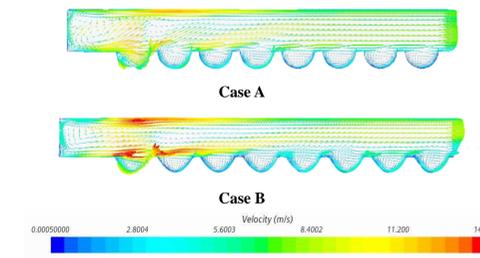


Figure 6: Velocity distribution along the leg-2 at Reynolds number = 20,000 (Side view)

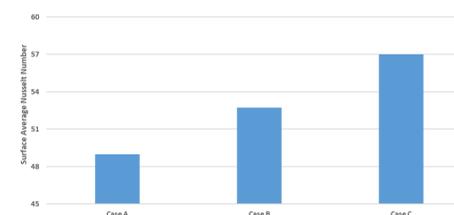


Figure 7: Surface average Nusselt number at Reynolds number = 20,000

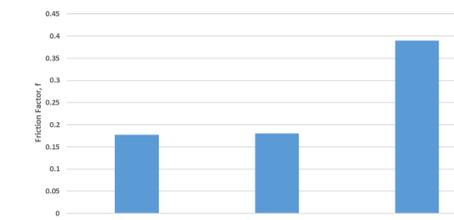


Figure 8: Friction factor at Reynolds number = 20,000

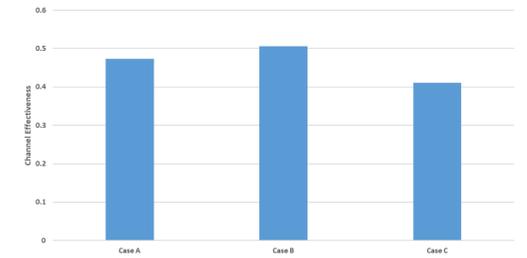


Figure 9: Channel effectiveness at Reynolds number = 20,000

Conclusions

- ❖ Computational Fluid Dynamics was used to investigate the heat transfer phenomena in the dimpled channel at Reynolds number=20,000.
- ❖ Three cases were studied-
 - Case A-Hemispherical Dimpled Channel
 - Case B- Leaf Dimpled Channel
 - Case C-Hemispherical Dimpled Channel without Bend
- ❖ After mesh independence study, 6.2 million cells were selected with surface remesher, prism layer and trimmer. $k-\omega$ SST model was used to solve the heat transfer model with air as a coolant.
- ❖ Heat transfer is the highest in Case C as it experiences the combination of impingement and cross-flow. The strong air jet impingement contributes a better heat transfer distribution in the second leg even though the total holes' area is less than the bend region.
- ❖ Case A and B experience flow attachment, vortex rolls in the downstream. The flow pattern explains the better heat transfer phenomena of the leaf dimpled channel than the hemispherical one.
- ❖ Even though the heat transfer enhancement is the highest in Case C, it is less effective than the other two channels due to excessive pressure drop. Case B shows the best performance in terms of heat transfer, pressure drop and effectiveness.

Literature cited

- Rao, Y., Feng, Y., Li, B. and Weigand, B., 2015. Experimental and numerical study of heat transfer and flow friction in channels with dimples of different shapes. *Journal of Heat Transfer*, 137(3), p.031901
- Xie, G., Sundén, B. and Zhang, W., 2011. Comparisons of pins/dimples/protrusions cooling concepts for a turbine blade tip-wall at high Reynolds numbers. *Journal of Heat Transfer*, 133(6), p.061902.

- Shen, Z., Qu, H., Zhang, D. and Xie, Y., 2013. Effect of bleed hole on flow and heat transfer performance of U-shaped channel with dimple structure. *International Journal of Heat and Mass Transfer*, 66, pp.10-22

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For further information

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